
Module-4

Lecture-20

Maneuvering Flight: Stick Fixed Maneuvering point

Stick fixed maneuvering point

Stick fixed maneuvering point is the *c.g.* location at which $d\delta e/dn = 0$

$$\frac{d\delta e}{dn} = -\frac{1}{V^2} \left[\frac{1.1gl_t}{\tau} \left(1 + \frac{k'}{n^2} \right) + \left\{ \frac{\frac{2W}{S}}{\rho C_{m_{\delta e}}} \left[\frac{dC_m}{dC_L} \right]_{fix} \right\} \right]$$

$$\frac{d\delta e}{dn} = -\frac{1}{V^2} \frac{2 \left(\frac{W}{S} \right)}{\rho C_{m_{\delta e}}} \left[\frac{1.1gl_t \rho C_{m_{\delta e}}}{2\tau \left(\frac{W}{S} \right)} \left(1 + \frac{k'}{n^2} \right) + \bar{x}_{cg} - \bar{n}_o \right] \quad (1)$$

$$\bar{x}_{cg} - \bar{n}_o = -\frac{1.1gl_t \rho C_{m_{\delta e}}}{2\tau \left(\frac{W}{S} \right)} \left(1 + \frac{k'}{n^2} \right) \quad (2)$$

Pull-up; $k' = 0$

$$\bar{x}_{cg} - \bar{n}_o = -\frac{1.1gl_t \rho C_{m_{\delta e}}}{2\tau \left(\frac{W}{S} \right)}$$

$$\bar{x}_{cg} = \bar{n}_m$$

so Equation 2 becomes,

$$\bar{n}_o = \bar{n}_m + \frac{1.1gl_t \rho C_{m_{\delta e}}}{2\tau \left(\frac{W}{S} \right)} \left(1 + \frac{k'}{n^2} \right) \quad (3)$$

Put Equation 3 in 1 to get;

$$\frac{d\delta e}{dn} = -\frac{1}{V^2} \frac{2 \left(\frac{W}{S} \right)}{\rho C_{m_{\delta e}}} \left[\frac{1.1gl_t \rho C_{m_{\delta e}}}{2\tau \left(\frac{W}{S} \right)} \left(1 + \frac{k'}{n^2} \right) + \bar{x}_{cg} - \bar{n}_m - \frac{1.1gl_t \rho C_{m_{\delta e}}}{2\tau \left(\frac{W}{S} \right)} \left(1 + \frac{k'}{n^2} \right) \right]$$

$$\frac{d\delta e}{dn} = -\frac{1}{V^2} \frac{2 \left(\frac{W}{S} \right)}{\rho C_{m_{\delta e}}} (\bar{x}_{cg} - \bar{n}_m)$$

We know that stick fixed neutral point is that *c.g.* location at which $d\delta e/dn = 0$, equivalently at $\bar{x}_{cg} = \bar{n}_m$; $d\delta e/dn = 0$

Stick fixed maneuvering point - a closer look

$$\frac{d\delta e}{dn} = -\frac{1}{V^2} \frac{2 \left(\frac{W}{S} \right)}{\rho C_{m\delta e}} (\bar{x}_{cg} - \bar{n}_m)$$

$\bar{n}_m - \bar{x}_{cg}$ is called maneuvering margin

The significant points to be made about the above equation are:

- The derivative $d\delta e/dn$ varies with maneuver margin.
- For more forward c.g. location, more elevator will be required to obtain the limit load factor. Therefore, as the c.g. moves forward, more elevator deflection is necessary to obtain a given load factor.
- The lower positive speed (higher the C_L) more elevator will be necessary to set the limit load factor. Thus, at low speeds more elevator deflection is necessary to obtain a desired load factor than is required to obtain at a higher speed.
- The derivative $d\delta e/dn$ should be linear with respect to c.g. at a constant C_L .