

ME-662 CONVECTIVE HEAT AND MASS TRANSFER

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LECTURE-35 BOUNDARY LAYER FLOW MODEL

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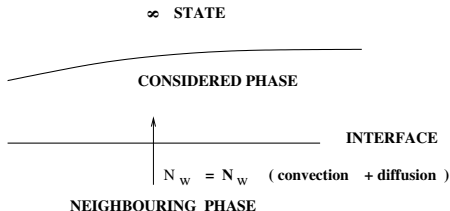
- 1 Definitions
- 2 Governing Equations
- 3 Conserved property eqns for all types of mass transfer
- 4 Boundary Conditions
 - 1 Mass Conservation Principle
 - 2 Energy Conservation Principle
- 5 $N_w = g \times B$ for small and large mass transfer rates

BL Flow Model - L35($\frac{1}{14}$)

- 1 Conv MT takes place due to **concentration gradients** of the transferred species
- 2 Since Reynolds flow model mimics the real flow, **Interface mass transfer flux** N_w ($\text{kg}/\text{m}^2\text{-s}$) from

$$N_w = g B$$

- 3 N_w and g have same units



N_w is **positive** when mass transfer takes place from the *neighbouring phase* into the *considered phase* across the interface & *vice versa*

Governing Equations - L35($\frac{2}{14}$)

Assuming **Steady-state mass transfer**

$$\frac{\partial(\rho_m u \Psi)}{\partial x} + \frac{\partial(\rho_m v \Psi)}{\partial y} = \frac{\partial}{\partial y} \left[\Gamma_\Psi \frac{\partial \Psi}{\partial y} \right] + S_\Psi$$

Ψ	Γ_Ψ	S_Ψ	
1	0	0	Bulk Mass
u	$\mu_{m,eff}$	$- dp / dx$	Momentum
ω_k	$\rho_m D_{eff}$	R_k	Species transfer
η_α	$\rho_m D_{eff}$	0	Element transfer
h_m	$k_{m,eff} / c p_m$	$-\partial(\sum m''_{y,k} h_k) / \partial y$	Energy

where $m''_{y,k} = -\rho_m D_{eff} \partial \omega_k / \partial y$. Sources Dp/Dt , \dot{Q}_{rad} , \dot{Q}_{others} and $\mu_{eff} (\partial u / \partial y)^2$ are ignored in the energy equation. **All equations are coupled requiring numerical solutions**. Simplifications of ω_k and h_m equations are possible under certain assumptions so that they are rendered to **conserved property equations**.

Conserved Property Eqn - L35($\frac{3}{14}$)

$$\frac{\partial(\rho_m u \Psi)}{\partial x} + \frac{\partial(\rho_m v \Psi)}{\partial y} = \frac{\partial}{\partial y} \left[\Gamma_\Psi \frac{\partial \Psi}{\partial y} \right]$$

$$N_w = g \times B \quad \text{with} \quad B = \frac{\Psi_\infty - \Psi_w}{\Psi_w - \Psi_T}$$

- 1 In **Inert MT without HT** , $\Psi = \omega_v$ and $\Gamma = \rho_m D$
- 2 In **Inert MT with HT** , $\Psi = \omega_v$ and h_m and $\Gamma_{mh} = \rho_m D = \rho_m \alpha_m$ with $Le = 1$
- 3 In **MT with SCR** , $\Psi =$ appropriate ϕ and h_m and $\Gamma_{mh} = \rho_m D = \rho_m \alpha_m$ with $Le = 1$ and equal $c_{p,k} = c_{pm}$
- 4 In **MT with ACR** , $\Psi =$ appropriate ϕ and $\Gamma_m = \rho_m D$

In each case, we need **Boundary Conditions** at $y = 0$ in w-state.

BCs - Mass Conservation - L35($\frac{4}{14}$)

- ① For **Inert Mass Transfer**, consider mass conservation between T- and w-states. Then

$$N_w \omega_{k,T} = N_w \omega_{k,w} - \rho_m D \frac{\partial \omega_k}{\partial y} \Big|_w$$

$$N_w = \frac{\rho_m D (\partial \omega_k / \partial y)_w}{\omega_{k,w} - \omega_{k,T}}$$

- ② For **Conserved property Φ**

$$N_w \Phi_T = N_w \Phi_w - \rho_m D \frac{\partial \Phi}{\partial y} \Big|_w$$

$$N_w = \frac{\rho_m D (\partial \Phi / \partial y)_w}{\Phi_w - \Phi_T}$$

where $\Phi = \omega_{fu} - \omega_{O_2} / r_{st} = \omega_{fu} + \omega_{pr} / (1 + r_{st})$ for an SCR
or $\Phi = \sum a_\alpha \eta_\alpha$ and a_α are **suitable chosen coefficients** for
an ACR

BCs - Energy Conservation - 1 - L35($\frac{5}{14}$)

- 1 Consider control volume between T- and w- states. Then

$$N_w h_{m,T} = N_w h_{m,w} + \left(\sum_k - \rho_m D \frac{\partial \omega_k}{\partial y} \Big|_w h_k \right) - q_w \quad \text{where}$$

$$q_w = k_m \frac{\partial T}{\partial y} \Big|_w = c_{pm} \Gamma_h \frac{\partial T}{\partial y} \Big|_w \quad \text{hence}$$

$$N_w = \frac{(\sum_k \Gamma_m (\partial \omega_k / \partial y)_w h_k) + c_{pm} \Gamma_h (\partial T / \partial y)_w}{h_{m,w} - h_{m,T}}$$

- 2 This is the general energy conservation principle. The **final form of the Numerator** will depend on mass transfer application.

BCs - Energy Conservation - 2 - L35($\frac{6}{14}$)

1 For Inert MT with HT , Le = 1 gives $\Gamma_h = \Gamma_m$. Hence

$$c_{pm} \Gamma_h \left(\frac{\partial T}{\partial y} \right)_w = \Gamma_h \left(\sum_k \omega_k c_{p,k} \right) \left(\frac{\partial T}{\partial y} \right)_w = \Gamma_h \left(\sum_k \omega_k \frac{\partial h_k}{\partial y} \right)_w$$

2 Hence,

$$\begin{aligned} N_w &= \frac{\Gamma_m \left(\sum_k (\partial \omega_k / \partial y)_w h_k \right) + \Gamma_h \left(\sum_k \omega_k \partial h_k / \partial y \right)_w}{h_{m,w} - h_{m,T}} \\ &= \frac{\Gamma_{mh} \left(\sum_k \{ \partial (\omega_k h_k) / \partial y \}_w \right)}{h_{m,w} - h_{m,T}} \\ N_w &= \frac{\Gamma_{mh} (\partial h_m / \partial y)_w}{h_{m,w} - h_{m,T}} \end{aligned}$$

BCs - Energy Conservation - 3 - L35($\frac{7}{14}$)

- ① For MT with HT and SCR, taking $Le = 1$, $c_{p,k} = c_{pm}$ and $\Delta T = (T - T_{ref})$, we have

$$h_{fu} = c_{pm} \Delta T + \omega_{fu} \Delta h_c \quad \text{and} \quad h_{O_2} = h_{pr} = c_{pm} \Delta T$$

- ② Hence,

$$\left(\sum_k \Gamma_m \left(\frac{\partial \omega_k}{\partial y} \right)_w h_k \right) = \Gamma_m \left(\frac{\partial \omega_{fu}}{\partial y} \right)_w \Delta h_c \quad \text{because}$$

$$c_{pm} \Delta T \Gamma_h \sum_k \left(\frac{\partial \omega_k}{\partial y} \right)_w = 0$$

$$\text{and} \quad c_{pm} \Gamma_h \left(\frac{\partial T}{\partial y} \right)_w = \Gamma_h \left(\frac{\partial h_m}{\partial y} \right)_w - \Gamma_h \Delta h_c \left(\frac{\partial \omega_{fu}}{\partial y} \right)_w$$

- ③ Hence, substitution with $\Gamma_m = \Gamma_h = \Gamma_{mh}$ gives

$$N_w = \frac{\Gamma_{mh} \left(\frac{\partial h_m}{\partial y} \right)_w}{h_{m,w} - h_{m,T}}$$

BCs - Energy Conservation - 4 - L35($\frac{8}{14}$)

- 1 Finally, for single component convective mass transfer

$$\left(\sum_k \Gamma_m \left(\frac{\partial \omega_k}{\partial y} \right)_w h_k \right) = 0 \quad \text{because } \omega_k = 1$$

$$\text{and } c_{pm} \Gamma_h \left(\frac{\partial T}{\partial y} \right)_w = \Gamma_h \left(\frac{\partial h_m}{\partial y} \right)_w$$

- 2 Hence

$$N_w = \frac{\Gamma_h (\partial h_m / \partial y)_w}{h_{m,w} - h_{m,T}}$$

- 3 If specific heats in all states are equal

$$N_w = \frac{\Gamma_h (\partial T / \partial y)_w}{T_w - T_T}$$

Comments - 1 - L35($\frac{9}{14}$)

- 1 Thus in all cases of mass transfer, **mass and energy conservation principles** give identical formula for N_w
- 2 Combining with Reynolds flow model which claims to mimic the real boundary layer flow model, we have

$$N_w = \frac{\Gamma_\psi (\partial\psi/\partial y)_w}{\psi_w - \psi_T} = g \times \left[\frac{\psi_\infty - \psi_w}{\psi_w - \psi_T} \right] = (\rho_m V)_w$$

- 3 Hence,

$$N_w \propto (\psi_\infty - \psi_w) \propto \Gamma_\psi \left(\frac{\partial\psi}{\partial y} \right)_w \propto V_w$$

- 4 This shows that even when Γ is uniform, the ψ -eqn is non-linear because velocity field (u and v) is a function of V_w and $(\psi_\infty - \psi_w)$. This is akin to **Natural Convection** in which u and v are functions of $(T_\infty - T_w)$.

Comments - 2 - L35($\frac{10}{14}$)

- 1 In Natural convection, the momentum and energy eqns are coupled through buoyancy source in the momentum eqn . In contrast, in Mass transfer, momentum, energy and species eqns are coupled through boundary conditions .
- 2 The coupling between momentum and Ψ -eqns can be ignored when $N_w \propto V_w \rightarrow 0$. Thus

$$g^* \equiv \left(\frac{N_w}{B_\Psi} \right)_{N_w \rightarrow 0} = \frac{-\Gamma_\Psi (\partial \Psi / \partial y)_w}{\Psi_w - \Psi_\infty}$$

where g^* now depends only on the Ψ - profiles. This definition is analogous to that used to define heat transfer coefficient.

- 3 When N_w is large, coupling is strong and $N_w = g \times B_\Psi$. Hence, g must be a function of B_Ψ and $g^* \equiv g_{B_\Psi \rightarrow 0}$

Comments - 3 - L35($\frac{11}{14}$)

- ① By analysing **Experimental data** on mass transfer with and without combustion, Spalding¹ showed that **within experimental scatter** ,

$$\frac{g}{g^*} = \frac{N_w/B}{(N_w/B)_{N_w \rightarrow 0}} = F(B) \text{ only}$$

- ② This eqn shows that (g / g^*) is **not influenced by Re, Pr or Sc numbers** . The validity of this assertion will be examined later.
- ③ Thus, all that is required is **the value of g^* (evaluated from $h_{cof, V_w=0} / c_{pm}$)** and **$F (B)$** to obtain g .

¹Spalding D B Convective Mass Transfer, Edward Arnold Ltd, London (1963)

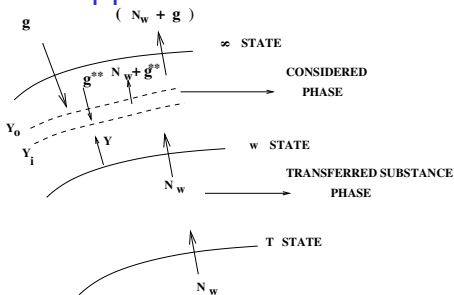
Form of $F(B) - 1 - L35(\frac{12}{14})$

- Using computer solutions of the BL eqns as well as experimental data, Spalding showed that

$$\frac{g}{g^*} = F(B) \simeq \frac{\ln(1+B)}{B}$$

This relationship was also predicted by the Stefan and Couette flow models.

- It can be derived for Reynolds flow model as well (see figure)



Consider 2 surfaces y_0 and y_i in the considered phase. Let **local Reynolds flux g^{**}** cross the y_0 surface carrying properties at y_0 . Similarly, let **Reynolds flux $g^{**} + N_w$** cross the y_i surface carrying properties at y_i .

Form of F (B) - 2 - L35($\frac{13}{14}$)

- 1 The physical idea behind introduction of g^{**} is that real flow processes like heat conduction, mass diffusion, turbulence etc do behave like the Reynolds flow but on a much smaller scale $\Delta y = (y_o - y_i) \rightarrow 0$.
- 2 Thus, writing mass conservation over y_o - and T-states

$$N_w \Psi_T + g^{**} \Psi_{y_o} = (g^{**} + N_w) \Psi_{y_i} \rightarrow \frac{N_w}{g^{**}} = \frac{\Psi_{y_o} - \Psi_{y_i}}{\Psi_{y_i} - \Psi_T} = \frac{d \Psi_y}{\Psi_{y_i} - \Psi_T}$$

- 3 Considering a large number of Δy between ∞ - and w-states

$$N_w \sum_w^{\infty} \frac{1}{g^{**}} = \int_0^{\infty} \frac{d \Psi_y}{\Psi_{y_i} - \Psi_T} = \ln \left[1 + \frac{\Psi_{\infty} - \Psi_w}{\Psi_w - \Psi_T} \right] = \ln(1 + B_{\Psi})$$

Form of $F(B)$ - 3 - L35($\frac{14}{14}$)

- 1 Thus, as $B_\psi \rightarrow 0$, $\sum_w^\infty (g^{**})^{-1} = B_\psi / N_w$
- 2 Therefore, comparison with the observation of slide 11 shows that as $B_\psi \rightarrow 0$, $\sum_w^\infty (g^{**})^{-1} = (g^*)^{-1}$. Hence,

$$N_w = g \times B_\psi = g^* \ln(1 + B_\psi) \text{ and}$$
$$\frac{g}{g^*} = F(B) = \frac{\ln(1 + B_\psi)}{B_\psi}$$

- 3 This formula can be used for large mass transfer rates obtained in liquid-fuel burning and transpiration cooling. Small mass transfer rates are encountered in Combustion of solid fuel or evaporative cooling/air-conditioning.
- 4 The validity of the formula will be checked in the next lecture.